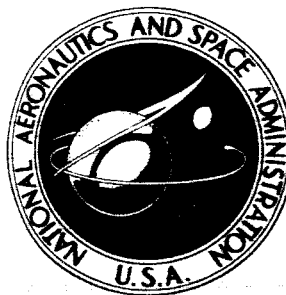


NASA TECHNICAL NOTE



NASA TN D-2840

NASA TN D-2840

FACILITY FORM 802

<b>N65-26257</b> (ACCESSION NUMBER)	
<u>38</u> (PAGES)	<u>1</u> (THRU)
	<u>33</u> (CODE)
	<u>33</u> (CATEGORY)
(NASA CR OR TMX OR AD NUMBER)	

GPO PRICE \$ \_\_\_\_\_  
*CRST1*  
DTB PRICE(S) \$ 2.00

Hard copy (HC) \_\_\_\_\_  
Microfiche (MF) .50

## COMPUTER PROGRAM DETAILS FOR DESIGN OF SENSIBLE-HEAT SPACE RADIATORS

*by Bruce M. Auer and Arthur V. Saule*

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**NATIONAL AERONAUTICS AND SPACE ADMINISTRATION**

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SUMMARY

26257

A program was designed for the calculation of the performance, weight, and area characteristics of a single-panel central-fin-tube flat-plate sensible-heat radiator for a set of thermodynamic and fluid mechanic conditions. The FORTRAN 7094 main program and its subroutines are reported herein. The program is based on the analysis, equations, and techniques developed and reported in NASA Technical Note D-2839.

A typical computer printout sheet is included and a discussion of the definitions of the FORTRAN symbols and groups of numbers that designate the inputs, the error tests, and the outputs is presented.

*Author*

INTRODUCTION

A design analysis and general characteristics of flat-plate central-fin-tube sensible-heat space radiators were presented in reference 1. It was shown therein that a combination of equations based on the relations of heat transfer, fluid mechanics, and meteoroid protection resulted in specific relations for calculating radiator geometry, weight, and panel planform area. In order to perform the parametric studies reported in reference 1, these equations and required inputs were programmed for an IBM 7094 electronic digital computer.

The purpose of this report is to describe the FORTRAN 7094 main program and its subroutines. Some phases of the computational procedure (e.g., Kalaba's method) will be discussed in more detail. Computer program diagrams are included to illustrate the procedure and to show how the various elements and subroutines are interconnected to generate the required outputs.

In addition, a typical printout sheet will be presented and explained. It contains the inputs and the outputs for two sensible-heat radiator examples: a Brayton cycle case and a secondary cooling radiator case. Various FORTRAN symbols and groups of quantities without symbols will be defined as they are written and appear on the printout sheet. The definitions correspond to the definitions of the symbols used in reference 1.

## SYMBOLS

A	area, sq ft
D	diameter, ft
F	angle factor
h	heat-transfer coefficient, Btu/(sec)(sq ft)(°R)
L	half-fin width, ft
Re	Reynolds number
R <sub>O</sub>	outside tube radius, ft
T	absolute temperature, °R
t	fin thickness, ft
U <sub>O</sub>	overall heat-transfer coefficient, Btu/(hr)(sq ft)(°R)
W	weight, lb
Z	tube length, ft
$\gamma_j$	heat-transfer parameter
$\delta_a$	armor thickness, ft
$\eta$	fin-tube effectiveness
$\theta$	dimensionless temperature ratio
$\lambda$	dimensionless conductance parameter

### Subscripts:

H	header
j	at location j
o	outside
s	sink
t	tube
tot	total
$\bar{X}$	at location $\bar{X}$

## COMPUTATIONAL PROCEDURE AND COMPUTER PROGRAM LISTING

### Procedure

As shown in the analysis section of reference 1, a combination of equations based on the relations of the heat transfer, fluid mechanics, and meteoroid protection resulted in certain relations for calculating radiator size, weight, and panel planform area. In order to perform the parametric studies, these equations and required inputs were programed for an IBM 7094 electronic digital computer

The FORTRAN 7094 program consists of two versions of the main program and four subroutines. The only difference in these versions is that the first version finds the fin-tube effectiveness by solving the proper equations using subroutines TOJ, DEQ2, and FXX. The second version using subroutine TABLE finds fin-tube effectiveness by interpolating in a table listing fin-tube effectiveness, which is read into the machine. Hereinafter, the main program refers to either version.

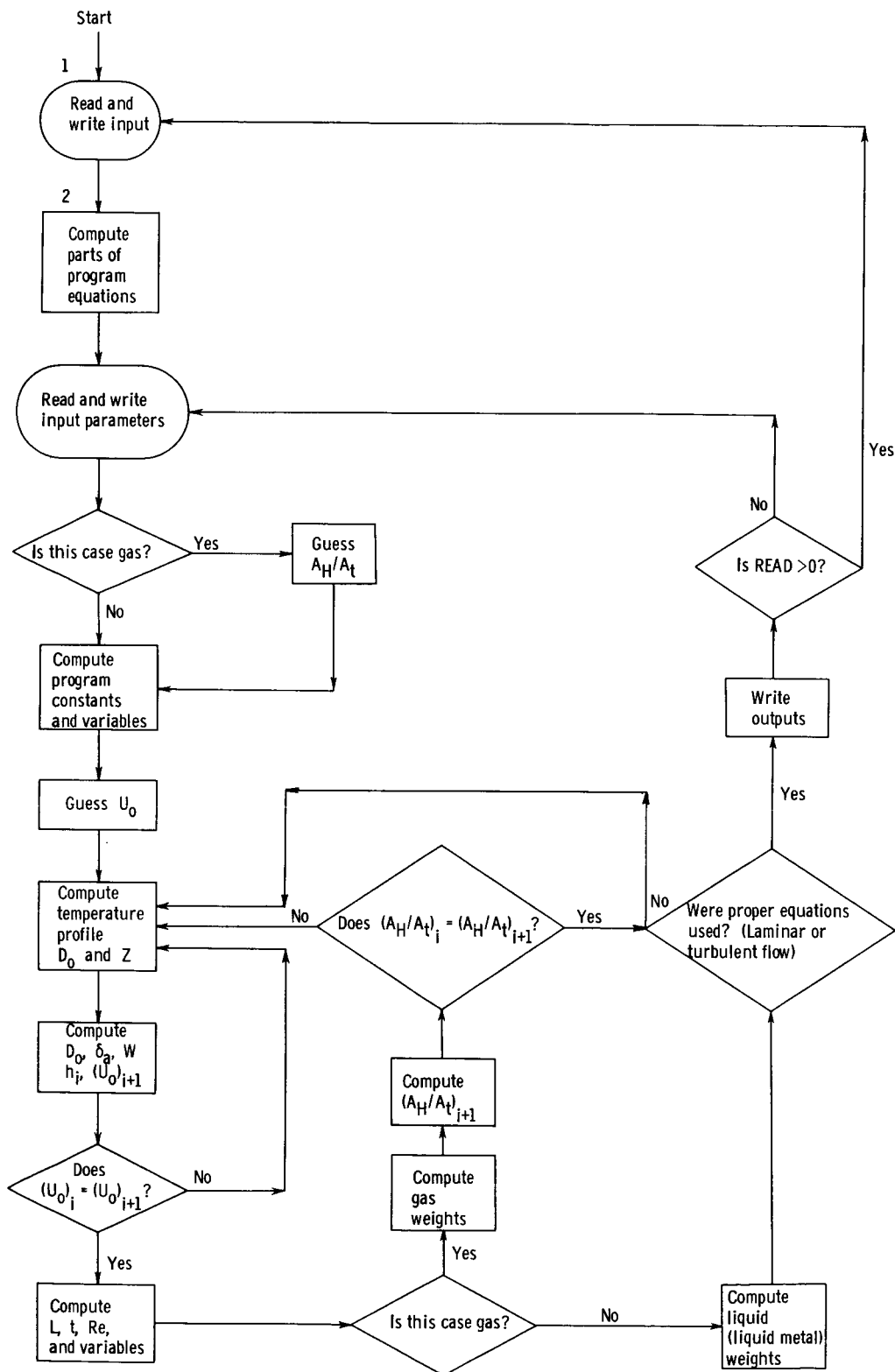
The flow diagram of the main program is shown in figure 1(a). The main program is written for either a gas, a liquid, or a liquid metal as the working fluid. For the radiator using gas as the working fluid, the program involves the ratio of header surface area to tube surface area  $A_H/A_t$ . For liquids and liquid metals the ratio  $A_H/A_t$  does not appear, since the header area is assumed to be negligible compared with the tube surface area.

Block 1 of figure 1(a) contains the main program inputs. They are the physical and thermal properties of the working fluid and the radiator structural materials (such as, sonic velocity, viscosity, specific heat, density, thermal conductivity, emissivity), as well as meteoroid protection parameters (meteoroid flux, density, operation time, probability of no meteoroid penetration, occlusion and spalling factors). Also included are the flow rate, inlet and outlet temperatures, the pressures and pressure drops, the accuracy percentages of the computations, the branching command numbers, and the number of points in the various meshes.

Parts of some equations can be computed based on the inputs and constants listed in the previous paragraph. This is done in block 2 of figure 1(a). The next step is to read and write the input parameters: inside tube diameter, fin-tube profile ratio, and fin conductance parameter at the radiator inlet. There are branches throughout the program that select either gas, liquid, or liquid metal radiator equations, variables, and constants. The program is written for tubes and headers with inside liners. Both turbulent-flow and laminar-flow equations are contained in the program for all three mediums: gas, liquid, and liquid metal. First, a solution is obtained for given inputs from turbulent-flow equations. After the solution is converged, the Reynolds number is checked. Then the Reynolds number is used as a test either to read the next set of inputs (if the Reynolds number indicates turbulent flow) or to find a second solution from the laminar-flow equations. A single result is printed out when the Reynolds number is less than 2300 or greater than 3000. If the Reynolds number is between 2300 and 3000, both laminar and turbulent flow results are printed.

The noteworthy feature of the main program is the iteration for the overall heat-transfer coefficient. The iteration starts with a guessed value and proceeds by using the method of false position (Regula Falsi) that is based on the principle that a curve over a short interval can be approximated by a straight line between two values, each being on opposite sides of the right value. The same method is used to calculate the ratio  $A_H/A_t$  for the gas working fluid. The overall heat-transfer coefficient iteration is accomplished within the  $A_H/A_t$  iteration.

During both iteration processes the values for the tube and header wall



(a) Main program.

Figure 1. - Computer program flow diagrams.

thickness, the tube outside diameter, the tube length, number of tubes, tube and liquid content weights, and the panel planform area are also obtained. With this information, the half-fin width, the fin thickness, fin weight, and header dimensions and weights are obtained.

Within the iteration of overall heat-transfer coefficient, the program solves for the tube surface temperature of each isothermal elemental strip by Newton's method. The first version requires a subroutine TOJ, the flow diagram for which is shown in figure 1(b). The fin-tube effectiveness is also computed by the subroutine TOJ. This involves solving the second-order differential equation

$$\left(\frac{d^2\theta}{d\bar{X}^2}\right) = \lambda \left[ \theta^4 - \theta_s^4 - (1 - \theta_s^4) F_{\bar{X}} \right] \quad (1a)$$

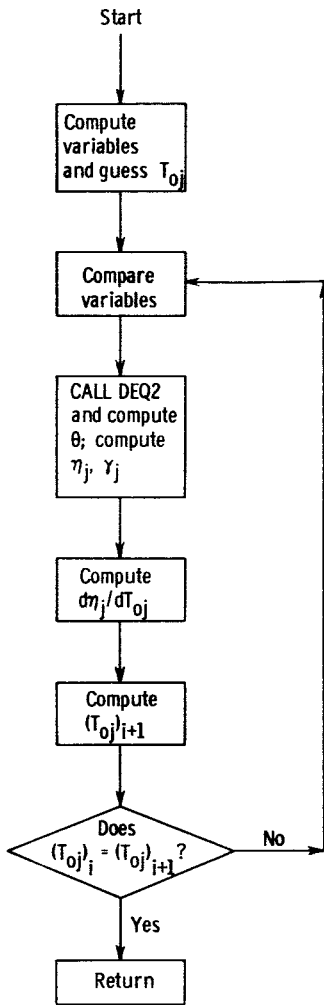
$$F_{\bar{X}} = F_{\bar{X}-1} + F_{\bar{X}-2} \quad (1b)$$

where

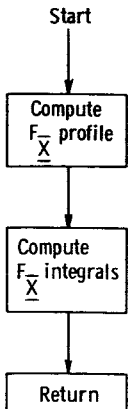
$$\left. \begin{aligned} \theta &= 1 \quad \text{at} \quad \bar{X} = 0 \\ d\theta/d\bar{X} &= 0 \quad \text{at} \quad \bar{X} = 1 \end{aligned} \right\} \quad (2)$$

by using subroutines DEQ2 and FXX. These additional subroutines will be discussed later. As part of Newton's method, the derivative  $d\eta$  is computed, and a value of  $T_{o,j}$  is obtained, which is checked against the guessed value depicted in the first block (fig. 1(b)). The iteration is repeated until the new value of  $T_{o,j}$  agrees with the previous value within the required accuracy.

The subroutine FXX (fig. 1(c)) computes the angle factors between fin and tube at each mesh point for use in the subroutine DEQ2 (fig. 1(d)). Subroutine FXX also obtains integrals independent of  $\theta$  to be used in subroutine TOJ (fig. 1(c)) for calculating the integral in the expression for the fin-tube effectiveness (ref. 1). The subroutine DEQ2 is used to solve the second-order differential equation (eqs. 1(a) and (b)) with its boundary conditions (eq. (2)). The method used in this program was devised by Kalaba (ref. 2). With this method,  $\theta^4$  is approximated by a linear function  $\theta$ , namely, by the first two terms of its Taylor series. The central differences are applied to  $d^2\theta/d\bar{X}^2$ . The linear approximation, the difference approximation to  $d^2\theta/d\bar{X}^2$ , and the boundary conditions are used to obtain a set of linear equations whose corresponding tri-diagonal matrix



(b) Subroutine TOJ.



(c) Subroutine FXX.

Figure 1 - Continued. Computer program flow diagrams.

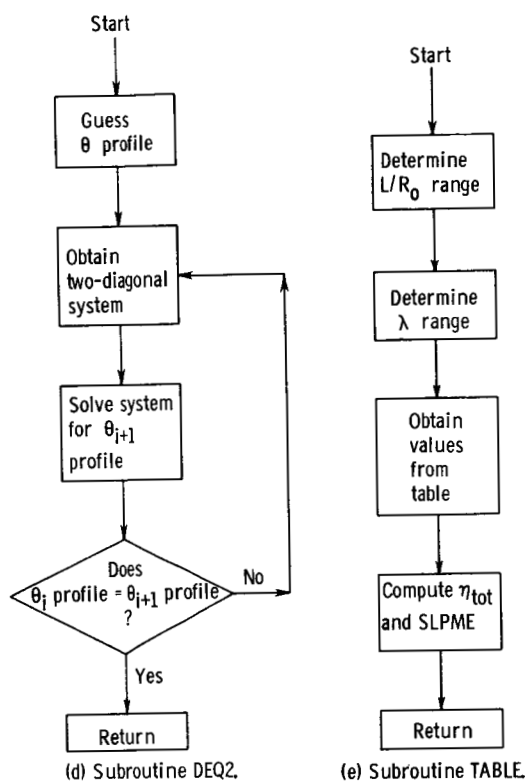


Figure 1. - Concluded. Computer program flow diagrams.

is reduced to two-diagonal form. The temperature  $\theta$  profile is then solved by backward substitution. An initial  $\theta$  profile guess is used and each successive iteration yields approximately one significant figure of accuracy.

By using the second version of the main program, computation time can be significantly reduced with the introduction of a table. The subroutine called TABLE (fig. 1(e)) now is used instead of subroutines TOJ, FXX, and DEQ2.

The values of the fin-tube effectiveness were first obtained from subroutines TOJ, FXX, and DEQ2 by using the fin-tube profile ratio, conductance parameter, and sink temperature ratio as independent variables. These values are then tabulated for use in subroutine TABLE. The first version may be used if these parameters are outside the range of TABLE or if it is desired to avoid the slight inaccuracy caused by using linear interpolation of nonlinear functions.



# Program Listings

```

$IBFTC GASLMC DECK
C      SAULE - GAS, LIQUID, LIQUID METAL RADIATOR                                0010
      DIMENSION TABL(13,36), TWR(100), DELZNW(100), DELZW(100),                0020
      1      FLWRIT(100) ,
      2      DIJ(20), FLRJ(20), FLAMRJ(20)
      COMMON TABL, FLR, FNC, ETATOT, SLPME, TUSE
      READ (5,130)((TABL(I,J), J=1,36), I=1,13)                                0060
130  FORMAT(12F6.3)                                                            0070
132  FORMAT(9F6.2)                                                            0090
      READ (5,134)PIE, FNMESH, FTEST, TEST, TESUH, FATST                      0100
134  FORMAT(E13.6, 6E10.3)                                                    0110
      WRITE (6,487)                                                            0120
215  READ (5,3)STEPH, ALPHA, BETA, EMACH, RHAP, VAVG, PNAN, FCCLU, TDAY,        0130
      1EPSL, ATICK, GSRLQ, QLMR, TZWR
      NMESH= FNMESH                                                            0150
      3  FORMAT(6E12.5)                                                        0160
      4  FORMAT(8E15.5)                                                        0190
C                                          0200
C      BRANCH - GAS OR LIQUID, LIQUID METAL RADIATOR                        0210
C
C      IF GAS CASE, THEN GSRLQ = 0, AND QLMR =
C      IF LIQUID CASE, THEN GSRLQ = 1 AND QLMR = 0
C      IF LIQUID METAL CASE , THEN GSRLQ = 1 AND QLMR = 1
C
C                                          0220
      2 IF(GSRLQ)160,160,150
150  READ (5,9)SCVEL, FMAS, VISC, UHETRD, CPH, TINLT, TEXTIT, RHAIN, RHAF,      0240
      1  RHAH, RHAT, PDRPT, PDRPH, THERKF, TCGAS, TS, RHAC                    0250
      IF (QLMR) 664,664,665
664  WRITE (6,662)
      GO TO 667
665  WRITE (6,666)
667  WRITE (6,4)SCVEL, FMAS, VISC, UHETRD, CPH, TINLT, TEXTIT, RHAIN, RHAF,    0260
      1RHAH, RHAT, PDRPT, PDRPH, THERKF, TCGAS, TS, RHAC, STEPH, ALPHA, BETA,  0270
      2EMACH, RHAP, VAVG, PNAN, FCCLU, TDAY, EPSL, ATICK, FTEST, TEST, TESUH,  0280
      3FNMESH, GSRLQ, QLMR, TZWR                                              0290
419  TCLQ=TCGAS                                                                0350
      RHAL=RHAIN                                                                0360
155  FF=1.0                                                                    0370
      GO TO 165                                                                0380
160  READ (5,9)SCVEL, FMAS, VISC, UHETRD, CPH, TINLT, TEXTIT, RGAS, PINLT,    0390
      1  RHAF, RHAH, RHAT, PTUBT, PTUBH, THERKF, TCGAS, TS, RHAC
      WRITE (6,663)
      WRITE (6,4)SCVEL, FMAS, VISC, UHETRD, CPH, TINLT, TEXTIT, RGAS, PINLT,  0410
      1RHAF, RHAH, RHAT, PTUBT, PTUBH, THERKF, TCGAS, TS, RHAC, STEPH, ALPHA,
      2BETA, EMACH, RHAP, VAVG, PNAN, FCCLU, TDAY, EPSL, ATICK, FTEST, TEST,
      3TESUH, FNMESH, FATST, GSRLQ, QLMR, TZWR
      PDRPT=PINLT*PTUBT                                                        0450
      PDRPH=PINLT*PTUBH                                                        0460
      PEXIT=PINLT-(PDRPT+2.0*PDRPH)                                           0470
      RHAIN=PINLT/(RGAS*TINLT)                                                0480
163  RHAUT=PEXIT/(RGAS*TEXTIT)                                                0490
      TINTD=1.0/TINLT                                                         0500
      TEXTD=1.0/TEXTIT                                                         0510

```

TINTD2=TINTD**2	0520
TEXTD2=TEXTD**2	0530
FF2=TEXTD2-TINTD2	0540
FF3=TEXTD2*TEXTD-TINTD2*TINTD	0550
FF=1.5*FF2/(FF3*TINLT)	0560
C	
165 FLTN= 2.0	
C	
DELTA=(TINLT -TEXIT)/FNMESH	
TEMPRD=TINLT-DELTA/2.0	0580
HI1=RHAIN/VISC	0590
HI3=VISC*CPH/TCGAS	0600
IF(QLMR)204,204,201	0610
201 HI4=HI3**0.4	0620
HI5=0.625*TCGAS*HI4	0630
GO TO 206	0640
204 HI4=HI3**0.3	0650
HI5=0.023*TCGAS*HI4	0660
206 DELZN1=FMAS*CPH*DELTA	0670
DELZN2=3600.0*DELZN1/PIE	0680
FMAS2=FMAS*FMAS	0690
DELTA1=(62.45*RHAP/RHAT)**0.5	0700
DELTA2=(VAVG/SCVEL)**(2.0/3.0)	0710
DELTA3=(0.6747E-04/RHAP)**(1.0/3.0)	0720
BETAD=1.0/(3.0*BETA)	0730
DELTA4=-TDAY*ALPHA/ALOG(PNAN)	0740
DELTA5=2.0/(3.0*EMACH*BETA+2.0)	0750
DELTA6=PIE*DELTA4*DELTA5	0760
DELTA1=DELTA1*DELTA2*DELTA3	0770
5 DELTA2=2.0*ATICK*FCCLU*DELTA1	0780
FMAS23=20.3/FMAS	0790
ZBG3=RHAIN*PDRPT/FF	0800
ZBG4=ZBG3**(.50/14.0)	0810
ZBG5=(1.0/VISC)**(1.0/14.0)	0820
6 ZBG7=ZBG4*ZBG5	0830
VNE1=4.0*FMAS/(PIE*RHAIN)	0840
WT1=PIE*RHAT	0850
TS4=TS**4	0860
7 WF1=2.0*RHAF	0870
8 READ(5,524) JDI,(DIJ(J),J=1,JDI)	
READ(5,524) JFLR,(FLRJ(J),J=1,JFLR)	
READ(5,524) JBRU,(FLAMRJ(J),J=1,JBRU),READ	
DO 525 JB1=1,JBRU	
FLAMRD= FLAMRJ(JB1)	
DO 526 JB2=1,JFLR	
FLR= FLRJ(JB2)	
DO 120 JB3=1,JDI	
DI = DIJ(JB3)	
WRITE (6,10) DI, FLR,FLAMRD	
10 FORMAT(1H0, 5H DI=E13.6,3X,	
15H FLR=E13.6,3X,7HFLAMRD=E13.6)	0910
C	
C	
IF(RHAC)460,460,461	
460 DLCC=0.0	
GO TO 464	
461 DI04=.04*DI	
IF(DI04-.00125) 417,415,415	0310
415 DLCC=DI04	0320
GO TO 464	
417 DLCC=.00125	0340

C  
C

464 DLCC2= DLCC+DLCC  
JL23= 0  
LADTR= 2  
DEBUG DLCC,DI

C  
C

605 IF(GSRLQ) 306,306,310	
306 JAT=1	0940
JAT1=0	0950
JAT2=0	0960
JAT3=2	0970
JAT4=1	0980
JAT5=0	0990
JAT6=0	1000
AHT=0.3	1010
AHT1=0.3	1020
308 AHT2=0.3	1030
GO TO 312	1040
310 JAT=0	1050
312 GAM1=STEPH*FLTN*EPSL/PIE	1060
TLITT1=STEPH*FLTN*EPSL/THERKF	1070
FNC=FLAMRD	1080
CALL TABLE	1090
ETARD=ETATOT	1100
DI2=DI*DI	1110
ZBG6=DI**(12.0/7.0)	1120
GAM2=ETARD*(1.0+FLR)	1130
11 GAM3=GAM2*GAM1	1140
UHETP=UHETRD	1150
KCNT=0	1160
14 UHET=UHETP	1170
TEMP1=TEMPRD	1180
FLAM1=FLAMRD	1190
ETA1=ETARD	1200
GAMMA1=GAM3/UHET	1210
GAMSAV=GAMMA1	1220
KPP=0	1230
TDMY=TEMP1	1240
17 TDMY2=TDMY*TDMY	1250
TDMY4=TDMY2*TDMY2	1260
FNWT1=GAMMA1*(TDMY4-TS4)	1270
18 FNWT=FNWT1+TDMY-TEMP1	1280
IF(ABS(FNWT)-FTFST) 25,25,20	1290
20 IF(KPP-20) 23,23,21	1300
21 WRITE (6,22)	1310
22 FORMAT(1H0,32H TROUBLE-LOOK AT STATEMENT NO.20)	1320
GO TO 120	1330
23 KPP=KPP+1	1340
TDMY3=TDMY2*TDMY	1350
DNWT=4.0*GAMMA1*TDMY3+1.	1360
24 TDMY=TDMY-FNWT/DNWT	1370
GO TO 17	1380
25 TSAVE=TDMY	1390
TZERC=TDMY**3	1400
DELZN=DELZN2/(TEMP1-TSAVE)/UHET	1410
DZN=DELZN	1420
TWR(1)=TDMY	1430
DELZNW(1)=DELZN	1440
JEND=NMESH	1450

FLWRIT(1)= FLAMRD	
30 DO 55 J=2,JEND	1460
KNN=0	1470
TEMP2=TEMP1-DELT	1480
TSAVE3=TSAVE**3	1490
DFNTW1=3.0*GAMMA1*FLAM1	1500
DFNTW2=DFNTW1/(ETA1*TSAVE3)	1510
TDMY=TEMP2	1520
35 FLAM2=FLAM1*(TDMY/TSAVE)**3	1530
36 FNC=FLAM2	1540
CALL TABLE	1550
37 ETA2=ETATDT	1560
GAMMA2=GAMMA1*(ETA2/ETA1)	1570
TDMY2=TDMY*TDMY	1580
TDMY4=TDMY2*TDMY2	1590
FNTW1=GAMMA2*(TDMY4-TS4)	1600
39 FNTW=TEMP2-TDMY-FNTW1	1610
IF(ABS(FNTW)-TFST)45,45,40	1620
40 IF(KNN-20) 43,43,41	1630
41 WRITE (6,42)	1640
42 FORMAT(1H0,32H TROUBLE-LOOK AT STATEMENT NO.40)	1650
GO TO 120	1660
43 KNN=KNN+1	1670
TDMY3=TDMY2*TDMY	1680
DFNTW3=(TDMY4-TS4)*TDMY2	1690
DFNTWT=DFNTW2*DFNTW3*SLPME	1700
DFNTW4=4.0*GAMMA2*TDMY3	1710
DFNTW=- (1.0+DFNTW4+DFNTWT)	1720
44 TDMY=TDMY-FNTW/DFNTW	1730
GO TO 35	1740
45 TSAVE=TDMY	1750
TEMP1=TEMP2	1760
GAMMA1=GAMMA2	1770
FLAM1=FLAM2	1780
48 ETA1=ETA2	1790
DELZN=DELZN2/(TEMP1-TSAVE)/UHET	1800
50 DZN=DZN+DELZN	1810
TWR(J)=TDMY	1820
DELZNW(J)=DELZN	1830
FLWRIT(J)=FLAM2	
55 CONTINUE	1840
IF(JAT)57,57,320	1850
320 DLAHT=1.0+AHT	1860
DELT8=(DZN*DELT6*DLAHT)**BETAD	1870
GO TO 324	1880
57 DELT8=(FLTIN*DZN*DELT6/2.0)**BETAD	1890
324 DELTA=DELTA2*DELT8	1900
60 DIAT=DI+FLTIN*(DELTA+DLCC)	
67 ZN=DZN/DIAT	2030
C	
C	
IF(LADTR-1) 446,446,448	
446 IF(GSRLQ) 450,450,452	
450 ZBG1= 0.7903*ZN*PINLT*PINLT*PTUBT*DI**4	
ZBIG= SQRT(ZBG1/(FMAS*RGAS*TINLT*VISC*FF))	
GO TO 69	
452 ZBG1= 0.7903*ZN*RHAIN*PDRPT*DI**4	
ZBIG= SQRT(ZBG1/FMAS/VISC)	
GO TO 69	
C	
C	

448	ZBG1=FMAS23*ZN	
	ZBG2=ZBG1**(.9/14.0)	2050
68	ZBIG=ZBG2*ZBG6*ZBG7	2060
69	FLATN=ZN/ZBIG	2070
	DEBUG LAOTR,ZBG1,ZBIG,ZN,PDRPT,DI	
	NBIG=FLATN	2080
	BIGN=FLD0AT(NBIG)+1.0	2090
	IF(FLATN-BIGN+1.0) 71,71,73	2100
71	BIGN=BIGN-1.0	2110
73	VNE=VNE1/(BIGN*DI2)	2120
C	IF(LAOTR-1) 456,456,454	
456	HI= 4.36*TCGAS/DI	
	GO TO 458	
C		
454	IF(QLMR)212,212,210	
210	HI2=(HI1*VNE*DI)**0.4	2140
	GO TO 75	2150
212	HI2=(HI1*VNE*DI)**0.8	2160
75	HI=HI5*HI2/DI	2170
458	DIAT36=3600.0/DIAT	
	DEBUG HI,HI2,VNE,BIGN	
C	BRANCH AND COMPUTE UD	
C		
	IF(QLMR) 520,520,503	
503	DADDI= DIAT/DI	
	U001 = DADDI/(3600.0*HI)	
	U0LN= ALOG(DADDI)	
	U002= DIAT*U0LN/(THERKF+THERKF)	
	UHETP= 1.0/(U001+U002)	
	GO TO 522	
C		
520	UHETP=DIAT36*HI*DI	2190
522	DIFFU=UHETP-UHET	2200
	TESH=(UHETP+UHET)*TESUH/2.0	2210
77	ADIFFU=ABS(DIFFU)-ABS(TESH)	2220
	IF(ADIFFU)85,85,78	2230
78	KCNT=KCNT+1	2240
	IF(KCNT-20) 14,14,82	2250
82	WRITE (6,83)	2260
83	FORMAT(1H0,32H TROUBLE-LOOK AT STATEMENT NO.78)	2270
	GO TO 120	2280
85	FLEN=FLR*DIAT/2.0	2290
	FLEN2=FLEN*FLEN	2300
	FLDI=DIAT+2.0*FLEN	2310
	TLITT2=TLITT1*FLEN2/FLAMRD	2320
	TLITT=TLITT2*TZERC	2330
	TLITTW=12.0*TLITT	2340
	WF2=WF1*FLEN	2350
	WT2=ZN*WT1	2360
	IF(GSRLQ)445,445,435	
435	DCC=DI+DLCC+DLCC	2380
	DCC2=DCC*DCC	2390
	WCT1=PIE*ZN/4.0	2400
	WCT2=RHAT*(DIAT*DIAT-DCC2)	2410
	WCT3=RHAC*(DCC2-DI*DI)	2420
	WT=WCT1*(WCT2+WCT3)	2430
	GO TO 87	2440
445	WT= PIE*ZN*(RHAT*DELTA*(DI+DLCC2 +DELTA) +RHAC*DLCC*(DI+DLCC))	
	DEBUG WT,DLCC,DELTA,ZN	
87	WF=WF2*ZN*TLITT	2460

PANEL=ZN*DIAT*(1.0+FLR)	2470
REYN=VNE*DI*RHAIN/VISC	2480
169 IF(GSRLQ)180,180,170	2490
170 P124=1.0/24.0	2500
P524=5.0/24.0	2510
P924=9.0/24.0	2520
DH1= PDRPH*RHAIN	2530
DH2= BIGN*FLDI/DH1	2540
DH4= DH2**P524	2550
DH5=0.174*VISC**P124	2560
DH6=FMAS**P924	2570
DH=DH4*DH5*DH6	2580
WLQH2=BIGN*FLDI*RHAH	2590
455 WCH1=2.0*PIE*BIGN*FLDI	2650
WCH2=RHAH*DELTA*(DH+DELTA+DLCC+DLCC)	2660
WCH3=RHAC*DLCC*(DH+DLCC)	2670
WLQH=WCH1*(WCH2+WCH3)	2680
173 WLQI2=2.0*BIGN*FLDI	2720
WLQI3=WLQI2*DH*DH	2730
WLQI4=WLQI3+DI2*ZN	2740
WLQI=PIE*RHAIN*WLQI4/4.0	2750
WLQ2=WT+WF	2760
174 WLQ=WLQ2+WLQH+WLQI	2770
C	
C	
DEBUG LAOTR,REYN,VNE,DI,RHAIN,VISC,BIGN,JL23	
IF(JL23-1) 607,609,601	
609 IF(REYN-2300.0) 648,610,610	
648 WRITE (6,646)	
GO TO 601	
610 WRITE(6,612) REYN	
GO TO 120	
607 IF(REYN-2300.0) 602,614,614	
602 JL23= 1	
LAOTR= 1	
WRITE (6,624) REYN	
GO TO 605	
614 JL23= 2	
C	
C	
601 IF(TZWR) 477,477,485	
477 DIAFLT=1.0/(DIAT*FLATN)	
DO 479 J=1,JEND	2800
DELZW(J)=DELZNV(J)*DIAFLT	2810
479 CONTINUE	2820
WRITE (6,467)(TWR(J),J=1,JEND)	2830
WRITE (6,469)(DELZW(J),J=1,JEND)	2840
485 WRITE (6,175)WLQ,WLQH,WLQI,WT,WF,DH,DIAT,PANEL,UHETP, DELTA,BIGN,	2850
1ZBIG,ZN,HI,VNE,TLITW,REYN	
175 FORMAT(1H ,5H WLQ=E13.6,3X,5HWLQH=E13.6,3X,5HWLQI=E13.6,3X,	2870
15H WT=E13.6,3X,5H WF=E13.6,3X,5H DH=E13.6/1X,5HDIAT=E13.6,3X,	2880
26HPANEL=E13.6,2X,6HUHETP=E13.6,4X,6HDFLTA=E13.6,2X,5HBIGN=E13.6,	2890
33X,5HZBIG=E13.6/1X,5H ZN=E13.6,2X,5H HI=E13.6,3X,5H VNE=E13.6,	2900
43X,6HTLITT=E13.6,3X,6H REYN=E13.6)	
C	
C	
IF(JL23-2) 120,613,120	
613 JL23= 3	
IF(REYN-3000.0) 616,120,120	
616 LAOTR= 1	
WRITE (6,646)	

GO TO 605	
180 FC1=PDRPH*RHAIN	2930
FC4=VISC/FMAS**0.6	2940
FC2=FLDI/FC1	2950
WH1=FMAS2/PDRPH	2960
WH5=(1.0/RHAIN)**0.2	2970
WH8=PIF*RHAH	3000
WH2=WH1*FLDI	3010
AH3=1.405*FLDI	3030
AT1=PIE*DIAT	3040
FC3=FC2*BIGN	3050
FC5=FC3**0.2	3060
FC6=FC5*FC4	3070
FSM=0.0382*FC6**0.208	3080
WH3=WH2*FSM*BIGN	3090
WH4=WH3**0.2	3100
DNE3=WH3/RHAIN	3210
DNE=0.596*DNE3**0.2	3220
DTW1=RHAIN/RHAUT	3230
DTW2=DTW1**0.2	3240
DTW=DTW2*DNE	3250
WHH= BIGN*FLDI	
WIH= PIE*WHH*((0.75*DNE +DLCC2 +DELTA)*DELTA*RHAH +(0.75*DNE+	
1 DLCC)*RHAC*DLCC)	
WDH=PIE*WHH*((0.75*DTW +DLCC2 +DELTA)*DELTA*RHAH	
1 (0.75*DTW+DLCC)*RHAC*DLCC)	
WH= WIH+WDH	
W= WH+WF+WT	
DEBUG WH,W,WDH,WHH,WIH,DELTA,BIGN	
C	
C	
AH1=WH4*WH5	3260
AH2=1.0+DTW2	3270
AH4=AH1*AH2*AH3	3280
95 AH=AH4*BIGN	3290
AT=AT1*ZN	3300
IF(JAT) 97,97,328	3310
328 KCNT=0	3320
IF(JAT6)330,330,395	3330
330 AHTC=AH/AT	3340
FAHT=AHT-AHTC	3350
IF(ABS(FAHT)/FAHT) 334,334,336	3360
334 NSG=-1	3370
GO TO 338	3380
336 NSG=1	3390
338 IF(JAT1) 340,340,341	3400
340 NSV1=NSG	3410
FAHTS1=FAHT	3420
AHTS1=AHT	3430
JAT1=1	3440
AHT1=0.25*AHT1	3450
AHT=AHT1	3460
GO TO 14	3470
341 IF(JAT2)342,342,360	3480
342 IF(NSG+NSV1)344,343,344	3490
343 JAT2=1	3500
FAHTS2=FAHT	3510
AHTS2=AHT	3520
GO TO 370	3530
344 IF(JAT4-6) 348,385,385	3540
348 JAT4=JAT4+1	3550

	GO TO (350,352),JAT3	3560
350	AHT1=0.25*AHT1	3570
	AHT=AHT1	3580
	JAT3=2	3590
	GO TO 354	3600
352	AHT2=AHT2+0.4	3610
	AHT=AHT2	3620
	JAT3=1	3630
354	UHETP=UHETRD	3640
	GO TO 380	3650
360	IF(NSV1+NSG)362,365,362	3660
362	NSV1=NSG	3670
	FAHTS1=FAHT	3680
	AHTS1=AHT	3690
	GO TO 370	3700
365	FAHTS2=FAHT	3710
	AHTS2=AHT	3720
370	AHTSV=AHT	3730
	AHTDS=(AHTS1-AHTS2)/((FAHTS1-FAHTS2)	3740
372	AHT=AHTS1-FAHTS1*AHTDS	3750
	ATST=0.5*(AHT+AHTSV)*FATST	3760
	IF(ABS(AHT-AHTSV)-ATST)390,390,38	3770
380	JAT5=JAT5+1	3780
	IF(JAT5-30)14,14,382	3790
382	WRITE (6,383)JAT5,AHTC	3800
	GO TO 395	3810
385	WRITE (6,387)JAT4,AHTC,NSV1	3820
	GO TO 395	3830
390	JAT6=1	3840
	GO TO 14	3850
	97 AHT=AH/AT	3860
C		
C	DEBUG REYN,VNE,DI,RHAIN,VISC,BIGN,JL23,LAOTR	
395	IF(JL23-1) 632,634,630	
634	IF(REYN-2300.0) 650,636,636	
650	WRITE (6,646)	
	GO TO 630	
636	WRITE(6,612) REYN	
	GO TO 120	
632	IF(REYN-2300.0) 638,640,640	
638	JL23= 1	
	LAOTR=1	
	WRITE(6,624) REYN	
	GO TO 605	
640	JL23= 2	
C		
C		
630	IF(TZWR)462,462,475	
462	DIAFLT=1.0/(DIAT*FLATN)	3880
	DO 465 J=1,JEND	3890
	DELZW(J)=DELZNW(J)*DIAFLT	3900
465	CONTINUE	3910
	WRITE (6,467)(TWR(J),J=1,JEND)	3920
	WRITE(6,469) (FLWRIT(J),J=1,JEND)	
	WRITE (6,469)(DELZW(J),J=1,JEND)	
475	WRITE (6,100)AH,DNE,DTW,BIGN,WT,Wf,WH,W,ZBIG,HI,AT,AHT,ZN,DELTA,	3930
	1DIAT,PANEL,UHETP,VNE,TLITTW,REYN	3940
C		
C		
	IF(JL23-2) 120,642,120	



```

642 JL23=3
    IF(REYN-3000.0) 644,120,120
644 LAOTR=1
    WRITE (6,646)
    GO TO 605
120 CONTINUE
526 CONTINUE
525 CONTINUE
530 IF(READ-1.0) 8,2,215

```

C  
C

```

100 FORMAT(1H ,5H AH=E13.6,3X,5H DNE=E13.6,3X,5H DTW=E13.6,3X,      3960
    15HBIGN=E13.6,3X,5H WT=E13.6,3X,5H WF=E13.6/1X,5H WH=E13.6,3X,    3970
    25H W=E13.6,3X,5HZBIG=E13.6,3X,5H HI=E13.6,3X,5H AT=E13.6,3X,    3980
    35H AHT=E13.6/1X,5H ZN=E13.6,2X,6HDELTA=E13.6,3X,5HDIAT=E13.6,    3990
    42X,6HPANEL=E13.6,2X,6HUHETP=E13.6,3X,5H VNE=E13.6/1X,          4000
    5 5HTLIT=E13.6,3X,5HREYN=E13.6)
383 FORMAT(1H ,28HTROUBLE - SEE STATE. NO. 382,4X,5HJAT5=I3,4X,      4060
    15HAHTC=E12.5)                                                    4070
387 FORMAT(1H ,28HTROUBLE - SEE STATE. NO. 344,4X,5HJAT4=I3,4X,      4080
    15HAHTC=E12.5,17)                                                4090
    9 FORMAT(11E7.5)                                                  4100
467 FORMAT(12F10.2)                                                  4110
469 FORMAT(12F10.4)                                                  4120
487 FORMAT (//,1H0,35H GAS, LIQUID, LIQUID METAL RADIATOR/
    1 ,1H0,73H NOTE - THE COMPUTED TEMP.,LAMDA AND DEL Z PROFIL      4130
    1ES ARE PRINTED OUT BELOW/
    2 1H0,65H THE OUTPUT IS FOR TURBULENT FLOW EQUAS., UNLESS OTHERWISE
    3 STATED)

```

C  
C

```

612 FORMAT(1H ,8HINSTABLE,4X,6HREYN =E12.5)
622 FORMAT(1H ,18HFOR LAMINAR EQUAS.,2X,6HREYN =E12.5)
624 FORMAT( 1H ,20HFOR TURBULENT EQUAS.,2X,6HREYN =E12.5)
646 FORMAT (1H0,31H OUTPUT FOR LAMINAR FLOW EQUAS.)
524 FORMAT (I4,(10E7.5))
662 FORMAT (1H0,/,20H INPUT DATA - LIQUID)
663 FORMAT (1H0,/,17H INPUT DATA - GAS)
666 FORMAT (1H0,/,26H INPUT DATA - LIQUID METAL)
END                                                                    4150

```

\$IBFTC TABLE DECK  
SUBROUTINE TABLE

C  
C

```

DIMENSION TABL(13,36)
COMMON TABL,FLR,FNC,ETATOT,SLPME,TUSE

```

C  
C  
C

```

TUSE= 0.0
6 FLSAVE=FLR
13 IF(FNC)79,18,18
18 IF(FLR-10.0)20,100,100
20 IF(FLR-8.0)310,301,301
310 IF(FLR-6.0)21,101,101
21 IF(FLR-4.0)22,102,102
22 IF(FLR-3.0)311,302,302

```

```

311 IF(FLR-2.0)23,103,103
  23 IF(FLR-1.5)312,303,303
312 IF(FLR-1.0)71,104,104
104 I=1
  DELL=0.5
  FLRT=1.0
  GO TO 60
103 I=3
  DELL=1.0
  FLRT=2.0
  GO TO 60
102 I=5
  DELL=2.0
  FLRT=4.0
  GO TO 60
101 I=6
  DELL=2.0
  FLRT=6.0
  GO TO 60
303 I=2
  FLRT=1.5
  DELL=0.5
  GO TO 60
302 I=4
  FLRT=3.0
  DELL=1.0
  GO TO 60
301 I=7
  FLRT=8.0
  DELL=2.0
  GO TO 60
100 IF(FLR-15.)299,299,300
300 IF(FLR-20.0)99,99,30
  30 IF(FLR-30.0)98,98,31
  31 IF(FLR-50.0)97,97,32
299 I=8
  FLRT=10.0
  DELL=5.0
  GO TO 60
99 I=9
  DELL=5.0
  FLRT=15.0
  GO TO 60
98 I=10
  DELL=10.0
  FLRT=20.0
  GO TO 60
97 I=11
  DELL=20.0
  FLRT=30.0
  GO TO 60
32 I=12
  IF(FLR-1050.0)40,41,41
41 FLR=1050
  GO TO 40
40 DELL=1000.0
  FLRT=50.0
  GO TO 60
60 IF(FNC-5.0)67,70,200
200 IF(FNC-20.0)209,211,206
206 IF(FLR-50.0)370,370,371

```

```

371 ETA=.145
    GO TO 372
370 ETA=.86567785+FLR*(-.1701984+FLR*(.012712537-FLR*.000191947))
372 WRITE (6,110)FNC
110 FORMAT(1H0,42HFNC OUT OF RANGE-SEE SUBROUTINE TABLE-FNC=E12.5)
    GO TO 71
209 FJ=21.+(FNC-5.0)
    J=FJ
    FAJ=J
    FRAC=FJ-FAJ
    K=2
    GO TO 61
67 FJ=FNC/0.25+1.0
    J=FJ
    FAJ=J
    FRAC=FJ-FAJ
    K=0
61 DEL1=TABL(I,J)-TABL(I+1,J)
    DEL2=TABL(I,J+1)-TABL(I+1,J+1)
    FRAC1=(FLR-FLRT)/DELL
    ETA1=TABL(I,J)-DEL1*FRAC1
    ETA2=TABL(I,J+1)-DEL2*FRAC1
    ETA=ETA1-ETA2
    IF(K-1)62,72,210
210 IF(K-2)213,213,214
62 ETATOT=ETA1-FRAC*ETA
    SLPME=-4.0*ETA
    FLR=FLSAVE
    GO TO 71
70 J=20
    K=1
    GO TO 61
72 ETATOT=ETA2
    SLPME=-4.0*ETA
    FLR=FLSAVE
    GO TO 71
213 ETATOT=ETA1-FRAC*ETA
    SLPME=-ETA
    FLR=FLSAVE
    GO TO 71
214 ETATOT=ETA2
    SLPME=-ETA
    FLR=FLSAVE
    GO TO 71
211 J=35
    K=3
    GO TO 61

```

```

C
C
C      TO SEE IF THE POINT (FLR,FNC) IS TABULATED IN THE TABLE
C      TUSE = 1, THEN POINT IS TABULATED
C      TUSE = 0, THEN POINT IS NOT TABULATED
C
C
C

```

```

71 IF (FLR-1.0) 424,401,401
401 IF (FLR-50.0) 402,402,424
402 IF (FRAC1) 424,410,403
403 IF (FRAC1-1.0) 424,410,424
410 IF (FNC)424,411,411
411 IF (FNC-20.0) 412,412,424

```

```
412 IF (FRAC) 413,423,413
413 IF (FNC-5.0) 424,423,414
414 IF(FNC-20.0) 424,423,424
```

```
C
C
```

```
423 TUSE= 1.0
      GO TO 425
424 TUSE= 0.0
425 RETURN
```

```
C
C
C
```

```
79 ETA=1.0
   RETURN
   END
```

\$IBFTC GASLM	LIST,REF,DECK	
C	SAULE - GAS,LIQUID,LIQUID METAL RADIATOR	0010
	DIMENSION TWR(100),DELZNW(100),DELZW(100),BCDMY(12),FXEQ(1000),	
1	FXAB(100)	
	COMMON FXEQ,FXAB,FLR,FXGAB,JINTG,MESH,TST,DTST,TS,TEMP2,FLAMTY,	
1	FLAMC,ETAA1,GAMA1,ETAA2,GAMA2,FLAM,FWRITE,FWRTSV	
	READ (5,490) (BCDMY(J),J=1,12)	
	READ (5,134) FNMESH,FMESH,TESUH,FATST,FJINTG,TST,DTST,FWRITE	
	WRITE(6,490) (BCDMY(J),J=1,12)	
	WRITE (6,487)	0120
215	READ (5,3)STEPH,ALPHA,BETA,EMACH,RHAP,VAVG,PNAN, FCCLU,TDAY,	0130
	1EPSL,ATICK,GSRLQ,QLMR,TZWR	0140
	NMESH=FNMesh	0150
2	PIE= 3.1415926	
	FWRTSV= FWRITE	
	JINTG= FJINTG	
	MESH= FMESH	
C		0200
C	BRANCH - GAS OR LIQUID,LIQUID METAL RADIATOR	0210
C		0220
	IF(GSRLQ)160,160,150	0230
150	READ (5,9)SCVEL,FMAS,VISC,UHETRD,CPH,TINLT,TEXIT,RHAIN,RHAF,	0240
1	RHAH,RHAT,PDRPT,PDRPH,THERKF,TCGAS,TS,RHAC	0250
	IF (QLMR) 664,664,665	
664	WRITE (6,662)	
	GO TO 667	
665	WRITE (6,666)	
667	WRITE (6,4)SCVEL,FMAS,VISC,UHETRD,CPH,TINLT,TEXIT, RHAIN,RHAF,	0260
	1RHAH,RHAT,PDRPT,PDRPH,THERKF,TCGAS,TS,RHAC,STEPH, ALPHA,BETA,	0270
	2EMACH,RHAP,VAVG,PNAN,FCCLU,TDAY,EPSL,ATICK,FNMESH,FMESH,	
3	TESUH,FATST,FJINTG,TST,DTST,FWRITE,GSRLQ,QLMR,TZWR	
419	TCLG=TCGAS	0350
	RHAL=RHAIN	0360
155	FF=1.0	0370
	GO TO 165	0380
160	READ (5,9)SCVEL,FMAS,VISC,UHETRD,CPH,TINLT,TEXIT, RGAS,PINLT,	0390
1	RHAF,RHAH,RHAT,PTUBT,PTUBH,THERKF,TCGAS,TS,RHAC	
	WRITE (6,663)	
	WRITE (6,4)SCVEL,FMAS,VISC,UHETRD,CPH,TINLT,TEXIT, RGAS,PINLT,	0410
	1RHAF,RHAH,RHAT,PTUBT,PTUBH,THERKF,TCGAS,TS,RHAC,STEPH,ALPHA,BETA,	
	1EMACH,RHAP,VAVG,PNAN,FCCLU,TDAY,EPSL,ATICK,FNMESH,FMESH,TESUH,	
3	FATST,FJINTG,TST,DTST,FWRITE,GSRLQ,QLMR,TZWR	
	PDRPT=PINLT*PTUBT	0450
	PDRPH=PINLT*PTUBH	0460
	PEXIT=PINLT-(PDRPT+2.0*PDRPH)	0470
	RHAIN=PINLT/(RGAS*TINLT)	0480
163	RHAUT=PEXIT/(RGAS*TEXIT)	0490
	TINTD=1.0/TINLT	0500
	TEXTD=1.0/TEXIT	0510
	TINTD2=TINTD**2	0520
	TEXTD2=TEXTD**2	0530
	FF2=TEXTD2-TINTD2	0540
	FF3=TEXTD2*TEXTD-TINTD2*TINTD	0550

	FF=1.5*FF2/(FF3*TINLT)	0560
C		
	165 FLTN= 2.0	
C		
	DELT=(TINLT -TEXIT)/FNMESH	0580
	TEMPRD=TINLT-DELT/2.0	0590
	HI1=RHAIN/VISC	0600
	HI3=VISC*CPH/TCGAS	0610
	IF(QLMR)204,204,201	0620
201	HI4=HI3**0.4	0630
	HI5=0.625*TCGAS*HI4	0640
	GO TO 206	0650
204	HI4=HI3**0.3	0660
	HI5=0.023*TCGAS*HI4	0670
206	DELZN1=FMAS*CPH*DELT	0680
	DELZN2=3600.0*DELZN1/PIE	0690
	FMAS2=FMAS*FMAS	0700
	DELT1=(62.45*RHAP/RHAT)**0.5	0710
	DELT2=(VAVG/SCVEL)**(2.0/3.0)	0720
	DELT3=(0.6747E-04/RHAP)**(1.0/3.0)	0730
	BETAD=1.0/(3.0*BETA)	0740
	DELT4=-TDAY*ALPHA/ALOG(PNAN)	0750
	DELT5=2.0/(3.0*EMACH*BETA+2.0)	0760
	DELT6=PIE*DELT4*DELT5	0770
	DELTA1=DELT1*DELT2*DELT3	0780
5	DELTA2=2.0*ATICK*FCCLU*DELTA1	0790
	FMAS23=20.3/FMAS	0800
	ZBG3=RHAIN*PDRPT/FF	0810
	ZBG4=ZBG3**(5.0/14.0)	0820
	ZBG5=(1.0/VISC)**(1.0/14.0)	0830
6	ZBG7=ZBG4*ZBG5	0840
	VNE1=4.0*FMAS/(PIE*RHAIN)	0850
	WT1=PIE*RHAT	0860
	TS4=TS**4	0870
7	WF1=2.0*RHAF	
8	READ (5,9) DI,FLR,FLAMRD,READ	
	WRITE (6,10)DI,FLR,FLAMRD	
10	FORMAT(1H0,5H DI=E13.6,3X,	
	15H FLR=E13.6,3X,7HFLAMRD=E13.6)	0910
C		
C		
	IF(RHAC)460,460,461	
460	DLCC=0.0	
	GO TO 464	
461	DI04=.04*DI	
	IF(DI04-.00125) 417,415,415	0310
415	DLCC=DI04	0320
	GO TO 464	
417	DLCC=.00125	0340
C		
C		
	464 DLCC2= DLCC+DLCC	
	JL23= 0	
	LAOTR= 2	
605	IF(GSRLQ) 306,306,310	
306	JAT=1	
	JAT1=0	0950
	JAT2=0	0960
	JAT3=2	0970
	JAT4=1	0980
	JAT5=0	0990

JAT6=0	1000
AHT=0.3	1010
AHT1=0.3	1020
308 AHT2=0.3	1030
GO TO 312	1040
310 JAT=0	1050
312 SEP= 2.0*STEPH*EPSL/PIE	
CALL FXX(FXEQ,FXAB,FXGAB,FLR,MESH,JINTG,FWRITE)	
TLITT1= 2.0*STEPH*EPSL/THERKF	
DI2=DI*DI	1110
11 ZBG6=DI**(12.0/7.0)	
UHETP=UHETRD	1150
KCNT=0	1160
14 UHET=UHETP	1170
SEPLR= SEP*(1.0+FLR)/UHET	
DZN= 0.0	
FLAMC= FLAMRD	
GAMA1= SEPLR	
ETAA1= 1.	
FLAMTY= 0.0	
TEMP2= TEMPRD	
CALL TOJ(TDMY)	
25 TSAVE=TDMY	1390
TZERC=TDMY**3	1400
DELZN=DELZN2/(TEMPRD-TSAVE)/UHET	
DZN=DELZN	1420
TWR(1)=TDMY	1430
DELZNW(1)=DELZN	1440
JEND=NMESH	1450
FLAMC= 0.	
30 DO 55 J=2,JEND	1460
GAMA1= GAMA2	
ETAA1= ETAA2	
TEMP2= TEMP2-DELT	
FLAMTY= FLAM/TDMY**3	
CALL TOJ(TDMY)	
DELZN=DELZN2/(TEMP2-TDMY)/UHET	
50 DZN=DZN+DELZN	1810
TWR(J)=TDMY	1820
DELZNW(J)=DELZN	1830
55 CONTINUE	1840
IF(JAT)57,57,320	1850
320 DLAHT=1.0+AHT	1860
DELT8=(DZN*DELT6*DLAHT)**BETAD	1870
GO TO 324	1880
57 DELT8=(DZN*DELT6)**BETAD	
324 DELTA=DELTA2*DELT8	1900
60 DIAT=DI+2.0*(DELTA+DLCC)	
67 ZN=DZN/DIAT	2030
DEBUG DLCC,DI	
C	
C	
C	
IF(LADTR-1) 446,446,448	
446 IF(GSRLQ) 450,450,452	
450 ZBG1= 0.7903*ZN*PINLT*PINLT*PTUBT*DI**4	
ZBIG= SQRT(ZBG1/(FMAS*RGAS*TINLT*VISC*FF))	
GO TO 69	
452 ZBG1= 0.7903*ZN*RHAIN*PDRPT*DI**4	
ZBIG= SQRT(ZBG1/FMAS/VISC)	
GO TO 69	

448	ZBG1=FMAS23*ZN	
	ZBG2=ZBG1**(.9.0/14.0)	2050
68	ZBIG=ZBG2*ZBG6*ZBG7	2060
69	FLATN=ZN/ZBIG	2070
	NBIG=FLATN	2080
	BIGN=FLD0AT(NBIG)+1.0	2090
	IF(FLATN-BIGN+1.0) 71,71,73	2100
71	BIGN=BIGN-1.0	2110
73	VNE=VNE1/(BIGN*DI2)	2120
C		
C		
C		
	IF(LADTR-1) 456,456,454	
456	HI= 4.36*TCGAS/DI	
	GO TO 458	
C		
454	IF(QLMR)212,212,210	
210	HI2=(HI1*VNE*DI)**0.4	2140
	GO TO 75	2150
212	HI2=(HI1*VNE*DI)**0.8	2160
75	HI=HI5*HI2/DI	2170
458	DIAT36=3600.0/DIAT	
C		
C		
	IF(QLMR) 520,520,503	
503	DA0DI= DIAT/DI	
	U001 = DA0DI/(3600.0*HI)	
	U0LN= ALOG(DA0DI)	
	U002= DIAT*U0LN/(THERKF+THERKF)	
	UHETP= 1.0/(U001+U002)	
	GO TO 522	
C		
520	UHETP=DIAT36*HI*DI	2190
522	DIFFU=UHETP-UHET	2200
	TESH=(UHETP+UHET)*TESUH/2.0	2210
77	ADIFFU=ABS(DIFFU)-ABS(TESH)	2220
	IF(ADIFFU)85,85,78	2230
78	KCNT=KCNT+1	2240
	IF(KCNT-20) 14,14,82	2250
82	WRITE (6,83)	2260
83	FORMAT(1H0,32H TROUBLE-LOOK AT STATEMENT NO.78)	2270
	GO TO 120	2280
85	FLEN=FLR*DIAT/2.0	2290
	FLEN2=FLEN*FLEN	2300
	FLDI=DIAT+2.0*FLEN	2310
	TLITT2=TLITT1*FLEN2/FLAMRD	2320
	TLITT=TLITT2*TZER	2330
	TLITTW=12.0*TLITT	2340
	WF2=WF1*FLEN	2350
	WT2=ZN*WT1	2360
	IF(GSRLQ) 445,445,435	
435	DCC=DI+DLCC+DLCC	2380
	DCC2=DCC*DCC	2390
	WCT1=PIE*ZN/4.0	2400
	WCT2=RHAT*(DIAT*DIAT-DCC2)	2410
	WCT3=RHAC*(DCC2-DI*DI)	2420
	WT=WCT1*(WCT2+WCT3)	2430
	GO TO 87	2440
445	WT= PIE*ZN*(RHAT*DELTA*(DI+DLCC2 +DELTA) +RHAC*DLCC*(DI+DLCC))	
87	WF=WF2*ZN*TLITT	2460
	PANEL=ZN*DIAT*(1.0+FLR)	2470



REYN=VNE*DI*RHAIN/VISC	2480
169 IF(GSRLQ)180,180,170	2490
170 P124=1.0/24.0	2500
P524=5.0/24.0	2510
P924=9.0/24.0	2520
DH1= PDRPH*RHAIN	2530
DH2= BIGN*FLDI/DH1	2540
DH4= DH2**P524	2550
DH5=0.174*VISC**P124	2560
DH6=FMAS**P924	2570
DH=DH4*DH5*DH6	2580
WLQH2=BIGN*FLDI*RHAH	2590
455 WCH1=2.0*PIE*BIGN*FLDI	2650
WCH2=RHAH*DELTA*(DH+DELTA+DLCC+DLCC)	2660
WCH3=RHAC*DLCC*(DH+DLCC)	2670
WLQH=WCH1*(WCH2+WCH3)	2680
173 WLQI2=2.0*BIGN*FLDI	2720
WLQI3=WLQI2*DH*DH	2730
WLQI4=WLQI3+DI2*ZN	2740
WLQI=PIE*RHAIN*WLQI4/4.0	2750
WLQ2=WT+WF	2760
174 WLQ=WLQ2+WLQH+WLQI	2770
DEBUG WT,DLCC,DELTA,ZN	

C  
C

```

DEBUG LAQTR,REYN,VNE,DI,RHAIN,VISC,BIGN,JL23
IF(JL23-1) 607,609,601
609 IF(REYN-2300.0) 648,610,610
648 WRITE (6,646)
GO TO 601
610 WRITE(6,612) REYN
GO TO 120
607 IF(REYN-2300.0) 602,614,614
602 JL23= 1
LAQTR= 1
WRITE (6,624) REYN
GO TO 605
614 JL23= 2

```

C  
C

601 IF(TZWR) 477,477,485	
477 DIAFLT=1.0/(DIAT*FLATN)	
DO 479 J=1,JEND	2800
DELZW(J)=DELZNV(J)*DIAFLT	2810
479 CONTINUE	2820
WRITE (6,467)(TWR(J),J=1,JEND)	2830
WRITE (6,469)(DELZW(J),J=1,JEND)	2840
485 WRITE (6,175)WLQ,WLQH,WLQI,WT,WF,DH,DIAT,PANEL,UHETP, DELTA,BIGN,	2850
1ZBIG,ZN,HI,VNE,TLITTW,REYN	
175 FORMAT(1H ,5H WLQ=E13.6,3X,5HWLQH=E13.6,3X,5HWLQI=E13.6,3X,	2870
15H WT=E13.6,3X,5H WF=E13.6,3X,5H DH=E13.6/1X,5HDIAT=E13.6,3X,	2880
26HPANEL=E13.6,2X,6HUHETP=E13.6,4X,6HDELTA=E13.6,2X,5HBIGN=E13.6,	2890
33X,5HZBIG=E13.6/1X,5H ZN=E13.6,3X,5H HI=E13.6,3X,5H VNE=E13.6,	2900
4 3X,6HTLITT=E13.6,3X,6H REYN=E13.6)	

C  
C

```

IF(JL23-2) 120,613,120
613 JL23= 3
IF(REYN-3000.0) 616,120,120
616 LAQTR= 1
WRITE (6,646)

```

GO TO 605	
180 FC1=PDRPH*RHAIN	2930
FC4=VISC/FMAS**0.6	2940
FC2=FLDI/FC1	2950
WH1=FMAS2/PDRPH	2960
WH5=(1.0/RHAIN)**0.2	2970
WH8=PIE*RHAH	3000
WH2=WH1*FLDI	3010
AH3=1.405*FLDI	3030
AT1=PIE*DIAT	3040
FC3=FC2*BIGN	3050
FC5=FC3**0.2	3060
FC6=FC5*FC4	3070
FSM=0.0382*FC6**0.208	3080
WH3=WH2*FSM*BIGN	3090
WH4=WH3**0.2	3100
DNE3=WH3/RHAIN	3210
DNE=0.596*DNE3**0.2	3220
DTW1=RHAIN/RHAUT	3230
DTW2=DTW1**0.2	3240
DTW=DTW2*DNE	3250
WHH= BIGN*FLDI	
WIH= PIE*WHH*((0.75*DNE +DLCC2 +DELTA)*DELTA*RHAH +(0.75*DNE+	
1 DLCC)*RHAC*DLCC)	
WOH=PIE*WHH*((0.75*DTW +DLCC2 +DELTA)*DELTA*RHAH	
1 (0.75*DTW+DLCC)*RHAC*DLCC)	
WH= WIH+WOH	
W= WH+WF+WT	
AH1=WH4*WH5	3260
AH2=1.0+DTW2	3270
AH4=AH1*AH2*AH3	3280
95 AH=AH4*BIGN	3290
AT=AT1*ZN	3300
IF(JAT) 97,97,328	3310
328 KCNT=0	3320
IF(JAT6)330,330,395	3330
330 AHTC=AH/AT	3340
FAHT=AHT-AHTC	3350
IF(ABS(FAHT)/FAHT) 334,334,336	3360
334 NSG=-1	3370
GO TO 338	3380
336 NSG=1	3390
338 IF(JAT1) 340,340,341	3400
340 NSV1=NSG	3410
FAHTS1=FAHT	3420
AHTS1=AHT	3430
JAT1=1	3440
AHT1=0.25*AHT1	3450
AHT=AHT1	3460
GO TO 14	3470
341 IF(JAT2)342,342,360	3480
342 IF(NSG+NSV1)344,343,344	3490
343 JAT2=1	3500
FAHTS2=FAHT	3510
AHTS2=AHT	3520
GO TO 370	3530
344 IF(JAT4-6) 348,385,385	3540
348 JAT4=JAT4+1	3550
GO TO (350,352),JAT3	3560
350 AHT1=0.25*AHT1	3570
AHT=AHT1	3580

JAT3=2	3590
GO TO 354	3600
352 AHT2=AHT2+0.4	3610
AHT=AHT2	3620
JAT3=1	3630
354 UHETP=UHETRD	3640
GO TO 380	3650
360 IF(NSV1+NSG) 362,365,362	3660
362 NSV1=NSG	3670
FAHTS1=FAHT	3680
AHTS1=AHT	3690
GO TO 370	3700
365 FAHTS2=FAHT	3710
AHTS2=AHT	3720
370 AHTSV=AHT	3730
AHTDS=(AHTS1-AHTS2)/(FAHTS1-FAHTS2)	3740
372 AHT=AHTS1-FAHTS1*AHTDS	3750
ATST=0.5*(AHT+AHTSV)*FATST	3760
IF(ABS(AHT-AHTSV)-ATST) 390,390,38	3770
380 JAT5=JAT5+1	3780
IF(JAT5-30) 14,14,382	3790
382 WRITE (6,383) JAT5,AHTC	3800
GO TO 395	3810
385 WRITE (6,387) JAT4,AHTC,NSV1	3820
GO TO 395	3830
390 JAT6=1	3840
GO TO 14	3850
97 AHT=AH/AT	3860
DEBUG WH,W,WOH,WHH,WIH,DELTA,BIGN	

C  
C  
C

```

DEBUG REYN,VNE,DI,RHAIN,VISC,BIGN,JL23,LAOTR
395 IF(JL23-1) 632,634,630
634 IF(REYN-2300.0) 650,636,636
650 WRITE (6,646)
GO TO 630
636 WRITE(6,612) REYN
GO TO 120
632 IF(REYN-2300.0) 638,640,640
638 JL23= 1
LAOTR=1
WRITE(6,624) REYN
GO TO 605
640 JL23= 2

```

C  
C  
C

630 IF(TZWR) 462,462,475	
462 DIAFLT=1.0/(DIAT*FLATN)	3880
DO 465 J=1,JEND	3890
DELZW(J)=DELZNV(J)*DIAFLT	3900
465 CONTINUE	3910
WRITE (6,467) (TWR(J),J=1,JEND)	3920
WRITE (6,469) (DELZW(J),J=1,JEND)	3930
475 WRITE (6,100) AH,DNE,DTW,BIGN,WT,WF,WH,W,ZBIG,HI,AT,AHT,ZN,DELTA, 1DIAT,PANEL,UHETP,VNE,TLITTW,REYN	3940

C  
C

```

IF(JL23-2) 120,642,120
642 JL23=3

```

```

        IF(REYN-3000.0) 644,120,120
644  LAQTR=1
        WRITE (6,646)
        GO TO 605
C
C
120 IF(READ-1.0)8,2,215                                4040
134 FORMAT(7E10.3)
    3 FORMAT(6E12.5)                                    0160
    4 FORMAT(8E15.5)                                    0190
490 FORMAT(12A6)
487 FORMAT (//,1H0,35H GAS, LIQUID, LIQUID METAL RADIATOR/
    1      ,1H0,73H NOTE - THE COMPUTED TEMP.,LAMDA AND DEL Z PROFIL
    1ES ARE PRINTED OUT BELOW/
    2 1H0,65H THE OUTPUT IS FOR TURBULENT FLOW EQUAS., UNLESS OTHERWISE
    3 STATED)
383 FORMAT(1H ,28HTROUBLE - SEE STATE. NO. 382,4X,5HJAT5=I3,4X,
    15HAHTC=E12.5)                                     4060
387 FORMAT(1H ,28HTROUBLE - SEE STATE. NO. 344,4X,5HJAT4=I3,4X,
    15HAHTC=E12.5,I7)                                  4070
    9 FORMAT(11E7.5)                                    4080
467 FORMAT(12F10.2)                                    4090
469 FORMAT(12F10.4)                                    4100
100 FORMAT(1H ,5H AH=E13.6,3X,5H DNE=E13.6,3X,5H DTW=E13.6,3X,
    15HBIGN=E13.6,3X,5H WT=E13.6,3X,5H WF=E13.6/1X,5H WH=E13.6,3X,
    25H W=E13.6,3X,5HZBIG=E13.6,3X,5H HI=E13.6,3X,5H AT=E13.6,3X,
    35H AHT=E13.6/1X,5H ZN=E13.6,2X,6HDELTA=E13.6,3X,5HDIAT=E13.6,
    42X,6HPANEL=E13.6,2X,6HUHETP=E13.6,3X,5H VNE=E13.6/1X,
    5 5HTLIT=E13.6,3X,5HREYN=E13.6)                    4110
612 FORMAT(1H ,8HINSTABLE,4X,6HREYN =E12.5)           4120
622 FORMAT(1H ,18HFOR LAMINAR EQUAS.,2X,6HREYN =F12.5)
624 FORMAT( 1H ,20HFOR TURBULENT EQUAS.,2X,6HREYN =E12.5)
646 FORMAT (1H0,31H OUTPUT FOR LAMINAR FLOW EQUAS.)
662 FORMAT (1H0,/,20H INPUT DATA - LIQUID)
663 FORMAT (1H0,/,17H INPUT DATA - GAS)
666 FORMAT (1H0,/,26H INPUT DATA - LIQUID METAL)
        END
                                                                4150

$IBFTC TOJ      LIST,REF,DECK
        SUBROUTINE TOJ(TDMY)
C
C      FOR SAULE - GAS-L-LM
        DIMENSION FXEQ(1000),FXAB(100),FXA(1000),THA(1000),
    1      FXTHA(1000),FTGA(1000)
        COMMON FXEQ,FXAB,FLR,FXGAB,JINTG,MESH,TST,DTST,TSA,TEMP2,
    1      FLAMTY,FLAMC,ETAA1,GAMA1,ETAA2,GAMA2,
    2      FLAM,FWRITE,FWRTSV
        FWRTSV=FWRTSV-1.0
    5  FJINTG=JINTG
        MESH1= MESH+1
    7  FMESH=MESH
        DX= 1.0/FMESH
        FLR1= FLR+1.0
        KTDMY=0
    8  TDMY=TEMP2
C
C
C      BEGIN TDMY INTERATION

```

C  
C  
C  
C  
C

TO SOLVE THE DIFFERENTIAL EQUA. USING SUB. DEQ2  
AND SUB. FXX

```

10 TDMY2=TDY*TDY
   THSA=THSA/TDMY
   FLAM=FLAMTY*TDY**3+FLAMC
12 THSA3=THSA**3
   THSA4=THSA3*THSA
14 THSA5=1.0-THSA4
   FLAMA5=FLAM*THSA5
16 FLAMA4=FLAM*THSA4
   DO 30 J=1,MESH1
18 FXA(J)=-FLAMA5*FXEQ(J)-FLAMA4
30 CONTINUE
   IF(FLAM) 26,22,26
22 DO 25 J=1,MESH1
   THA(J)=1.
25 CONTINUE
   GO TO 28
26 CALL DEQ2(THA,FXA,FLAM,4.0,MESH,TST,FWRTSV)
28 SLPA=(THA(2)-1.0)/DX

```

C  
C  
C

TO COMPUTE ETA AND GAMMA

```

   IF(FWRTSV)33,33,130
130 WRITE(6,132) DX,FLR1,TDMY,TDMY2,THSA,FLAM,THSA3,THSA4,
   1 THSA5,FLAMA5,FLAMA4,SLPA
132 FORMAT(1H , 8H TDJ-133,(/8E15.5))
33 DO 35 J=1,MESH1
34 FXTHA(J)= FXEQ(J)*THA(J)**4
35 CONTINUE
   CALL FNTGRL(MESH1,DX,FXTHA,FTGA)
   ET1A=2.0-THSA4
37 ET2A= ET1A*FXGAB -FTGA(MESH1)
   IF(FLAM)38,39,38
39 SLPMA=0.0
   ETAA2= 1.
   GAMAD= 0.
   GO TO 40
38 SLPMA=SLPA/FLAM
43 ET3A= THSA5 +FLR*(ET2A-SLPMA)
   ETAA2= ET3A/THSA5/FLR1
40 GAMA2= GAMA1*ETAA2/ETAA1

```

C  
C  
C  
C  
C  
C

END ETA AND GAMA COMP.

TO COMPUTE DERIVATIVE OF ETA WRT TDMY

```

   IF(FLAM)44,62,44
44 DRTHA1=4.0*THSA3*THSA/TDMY2
   TAFLR= THSA5*FLR1
   DXX=1.0/FJINTG
   DXX4= 4.0*DXX
   JDEL=MESH/JINTG
   JMESH=-JDEL/2+1
   IF(FWRTSV)46,46,140
140 WRITE(6,142) ET1A,ET2A,ET3A,ETAA2,GAMA2,
   1 DRTHA1,TAFLR,DXX,DXX4,SLPMA,

```

```

1      (FXEQ(I),FXA(I),THA(I),FXTHA(I),
1      THA(I),FXTHA(I),I=98,101)
142  FORMAT(1H , 8H TOJ-142,(/8E15.5))
46  ETAAD3=0.
C
C      TO COMPUTE AN APPROXIMATION TO THE DERIVATIVE
C      OF THETA WRT. TDMY
C
      DO 55 J=1,JINTG
      JMESH=JMFESH+JDFL
      TBARA = THA(JMESH)
48  TBARA3 = TBARA**3
      DRTHA2 = DRTHA1*(DXX-FXAB(J))
50  DRTHA= -DRTHA2/(DXX4*TBARA3)
      ETAAD1=4.0*TBARA3*(DXX-FXAB(J))
52  ETAAD2= DRTHA*ETAAD1
54  ETAAD3= ETAAD3+ETAAD2
      IF(FWRTSV)55,55,150
150  WRITE(6,152) JMESH,TBARA,TBARA3,DRTHA2,
1      DRTHA,ETAAD1,ETAAD2,ETAAD3
152  FORMAT(1H , 8H TOJ-152,I5,(/8E15.5))
55  CONTINUE
      ETAAD4= DRTHA1*FLR1*(1.0-ETAAD2)
58  ETAAD = (ETAAD4+FLR*ETAAD3)/TAFLR
60  GAMAD= GAMA1*ETAAD/ETAAD1
C
C
C      END COMP. OF DERIVATIVE OF ETA WRT TDMY
C
C      TO SOLVE FOR TDMY BY NEWTONS METHOD
C
62  TDMY3= TDMY2*TDMY
      TDMY4= TDMY3*TDMY
      TDMY34=4.0*TDMY3
      FNWTA= TDMY4-TSA**4
64  FNWT = TDMY-TEMP2+GAMA2*FNWTA
      DFNWTA= GAMA2*TDMY34 + GAMAD*FNWTA
66  DFNWT= 1.0+DFNWTA
      TSAVE= TDMY
      TDMY= TDMY-FNWT/DFNWT
68  TDTST= 0.5*ABS(TSAVE+TDMY)*DTST
      IF(FWRTSV)70,70,160
160  WRITE(6,162) KTDMY,ETAAD4,ETAAD,GAMAD,
1      TDMY3,TDMY4,TDMY34,FNWTA,FNWT,DFNWTA,DFNWT,
2      TSAVE,TDMY,TDTST
162  FORMAT(1H , 8H TOJ-160,I5,(/8E15.5))
70  IF(ABS(TSAVE-TDMY)-TDTST)90,90,85
85  KTDMY= KTDMY+1
      IF(KTDMY-30)10,10,87
87  WRITE(6,88)TDMY,TDTST,KTDMY
88  FORMAT(1H ,16H TROUBLE SUB TOJ,4X,5HTDMY=E12.5,4X,E12.5,4X,I3)
90  RETURN
      END

$IBFTC FXX      DECK
      SUBROUTINE FXX(FXEQ,FXAB,FXGAB,FLR,MESH,JINTG)
C
C      FOR SAULE - GAS-L-LM AND SOLAR PROGRAMS
      DIMENSION FXEQ(1000),FXAB(100)

```

```

3 FMESH=MESH
  FJINTG=JINTG
5 DX=1.0/FMESH
  MESH1=MESH+1
  X=-DX
7 FLRD=1.0/FLR
  FLRDS= FLRD*FLPD
  FLRD2= FLRD+2.0
9 FLRD5=0.5*FLRD

```

```

      TO OBTAIN FUNCTION(X) AT EACH MESH POINT
      FOR USE IN SUB DEQ2

```

```

      DO 30 J=1,MESH1
        X=X+DX
16 FXA1=FLRD+X
        FXA2= FXA1*FXA1-FLRDS
        FXA= 1.0-SQRT(FXA2)/FXA1
        FXB1= FLRD2-X
18 FXB2= SQRT(FXB1*FXB1-FLRDS)
        FXB= 1.0-FXB2/FXB1
20 FXEQ(J)= 0.5*(FXA+FXB)
30 CONTINUE

```

```

      TO OBTAIN JINT INTEGRALS OF FUNCTION (X)
      INTEGRALS TO BE USED IN SUB TOJ

```

```

      X=0.0
      FXAB1=0.0
34 DX=1.0/FJINTG
      FXGB1=FLRD2/FLRD
      FXGB2=SQRT(FXGB1*FXGB1-1.0)
36 FXGB3=FXGB2-ATAN(FXGB2)
      DO 50 J=1,JINTG
        X=X+DX
        FXGA1=(FLRD+X)/FLRD
        FXGA2=SQRT(FXGA1*FXGA1-1.0)
38 FXGA3= FXGA2-ATAN(FXGA2)
        FXGB4=(FLRD2-X)/FLRD
        FXGB5=SQRT(FXGB4*FXGB4-1.0)
40 FXGB6=FXGB5-ATAN(FXGB5)
        FXGB7=FXGB6-FXGB3
        FXAB2=-FLRD5*(FXGA3-FXGB7)
        FXAB(J)=DX+FXAB2-FXAB1
42 FXAB1=FXAB2
50 CONTINUE

```

```

52 FXGAB=1.0+FXAB2
55 RETURN
END

```

```

$IBFTC DEQ2      DECK
      SUBROUTINE DEQ2(TH,FX,C,POW,MESH,TST)

```

```

      FOR SAULE - GAS-L-LM AND SOLAR PROGRAMS
      DIMENSION TH(1000),FX(1000),FD(1000),DG(1000),DGD(1000)

```

```

      FMESH=MESH
14  DX=1.0/FMESH
      DX2=DX*DX
      DGC=C*POW*DX2
16  PM1=POW-1.0
      FDC=(-PM1)*C*DX2
18  KCNT=0
      DO 20 J=1,MESH
          TH(J)=1.0
20  CONTINUE
22  DGD(1)=0.
      DG(1)=1.0
23  FD(1)=1.0

C
C          TO REDUCE MATRIX FROM TRI-DIAGONAL TO TWO DIAG.
C
      DO 27 J=2,MESH
          THPOW=TH(J)**POW
25  DG(J)= -DGD(J-1)-2.0-DGC*THPOW/TH(J)
          DGD(J)=1.0/DG(J)
          FD1=FDC*THPOW+DX2*FX(J)
26  FD(J)=FD1-FD(J-1)/DG(J-1)
27  CONTINUE
          DGD(MESH)=1.0/(DG(MESH)+1.0)
28  TH(MESH+1)=0.0
          MESH2=MESH+2
30  KTST=0

C
C          TO SOLVE FOR TH BY BACK SUBSTITUTION
C
      DO 35 JJ=2,MESH
          J=MESH2-JJ
          THSV=TH(J)
31  TH(J)=DGD(J)*(FD(J)-TH(J+1))
          IF(ABS(THSV-TH(J))-TST)35,35,32
32  KTST=KTST+1
35  CONTINUE

C
36  IF(KTST)43,43,37
37  KCNT=KCNT+1
      IF(KCNT-25)22,22,40
40  WRITE(6,41) KCNT,KTST
43  TH(MESH+1)=TH(MESH)
41  FORMAT(1H ,23H TROUBLE SEE SUBR. DEQ2,1X,5HKCNT=I3,4X,5HKTST=I3)
45  RETURN
      END

```



## COMPUTER PRINTOUT SHEET

### Description and Symbols

This section contains a printout sheet from the electronic digital computer for an argon gas and an ether (ET-378) example. The sheet is typical for the inputs and outputs discussed in the analysis section of reference 1.

The first group of numbers under the title Input Data represents the physical and thermal properties of the working fluid, radiator material, and meteoroid protection parameters. A brief explanation of these inputs follows for a gas case as they are read from left to right. The symbols preceding these terms represent these inputs in the computational procedure of the computer program listing.

First line:

SCVEL	sonic velocity, ft/sec
FMAS	total mass flow rate, lb/sec
VISC	viscosity of gas, lb/(ft)(sec)
UHETRD	first guess of overall heat-transfer coefficient, Btu/(hr)(sq ft)(°R)
CPH	specific heat of gas, Btu/(lb)(°R)
TINLT	inlet gas temperature, °R
TEXTIT	exit gas temperature, °R
RGAS	gas constant, ft-lb/(lb)(°R)

Second line:

PINLT	inlet gas pressure, lb/sq ft
RHAF	fin density, lb/cu ft
RHAH	header density, lb/cu ft
RHAT	tube density, lb/cu ft
PTUBT	pressure drop fraction in tubes
PTUBH	pressure drop fraction in each header
THERKF	thermal conductivity of fin, Btu/(hr)(ft)(°R)
TCGAS	thermal conductivity of gas, Btu/(sec)(ft)(°R)

Third line:

TS        equivalent sink temperature of space, °R  
RHAC     liner density, lb/cu ft  
STEPH    Stefan-Boltzmann constant  
ALPHA    constant in meteoroid mass distribution  
BETA     constant in meteoroid mass distribution  
EMACH    velocity ratio exponent in meteoroid protection equation  
RHAP     meteoroid density, lb/cu ft  
VAVG     average meteoroid velocity, ft/sec

Fourth line:

PNAN     probability of no meteoroid penetration  
FCCLU    orientation or occlusion factor  
TDAY     radiator mission time, days  
EPSL     emissivity of surface coating  
ATICK    finite plate thickness and spalling correction factor  
FTEST    error test in computation of surface temperature  
TEST     error test in computation of surface temperature  
TESUH    percent error in overall heat-transfer coefficient

Fifth line:

FMESH    number of elemental isothermal strips  
FATST    percent error in header-area to tube-area ratio  
GSRLQ    branch number (gas or liquid)  
QLMR     branch number (liquid or liquid metal)  
TZWR     control for printing of surface temperatures, conductance parameter,  
          and length of elemental strip profiles

The second group of input data is for a radiator that uses liquid ether (ET-378) as a working fluid. It is similar to the inputs of the gas example

except that instead of RGAS (gas constant), RHAIN (liquid density, lb/cu ft) is printed. The inlet gas pressure PINLT is not printed for the liquid example. Instead of PIUBT (pressure drop fraction in tubes) and PTUBH (pressure drop fraction in each header) PDRPT (pressure drop in tubes, lb/sq ft) and PDRPH (pressure drop in each header, lb/sq ft) are printed. The difference in inputs of branch numbers GSRLQ and QLMR should also be noted. For the gas example these inputs are 0. and 0.; for the liquid example, 0.10000E 01 and 0.; and for the liquid metal example (not printed), 0.10000E 01 and 0.10000E 01, respectively.

These sets of inputs are for a simplified computational procedure when tabulated relations among effectiveness, conductance parameter, fin-tube profile ratio, and sink temperature are used. When a functional relation is used, additional inputs are included between inputs FATST and GSRLQ.

FJINTG    used in approximation of integral in effectiveness computations  
TST        absolute error for temperature ratio,  $\theta_j$   
DTST       percent error in surface temperature  $T_{o,j}$   
FWRITE    control for debug printing

The group of numbers after the input data starts with parametric inputs.

DI        inside tube diameter, ft  
FLR        fin-tube profile ratio  
FLAMRD    initial conductance parameter

The next line represents the surface or base temperatures at the middle of each elemental strip (TWR(J)), followed by a line of corresponding conductance parameter (FLWRIT(J)) and length of each elemental strip (DELZW(J)).

For the gas example, the following outputs are listed on the printout sheet, reading from left to right:

AH        total header area, sq ft  
DNE        maximum inside diameter of inlet header, ft  
DTW        maximum inside diameter of outlet header, ft  
BIGN       number of tubes  
WT        tube weight, lb  
WF        fin weight, lb  
WH        header weight, lb

W	total radiator weight, lb
ZBIG	tube length, ft
HI	inside film heat-transfer coefficient, $\text{Btu}/(\text{sec})(\text{sq ft})(^{\circ}\text{R})$
AT	total tube area, sq ft
AHT	ratio of header area to tube area
ZN	total tube length, ft
DELTA	tube and header armor thickness, ft
DIAT	outside tube diameter, ft
PANEL	panel planform area, sq ft
UHETP	overall heat-transfer coefficient, $\text{Btu}/(\text{hr})(\text{sq ft})(^{\circ}\text{R})$
VNE	inlet fluid velocity, ft/sec
TLIT	fin thickness, in.
REYN	Reynolds number

For the liquid example, most of the symbols are the same as for the gas example. The outputs and their symbols, that follow, are not used in the gas example.

WLQ	total radiator weight, lb
WLQH	header weight, lb
WLQI	liquid content weight, lb
DH	header inside diameter, ft
TLITT	fin thickness, in.

Lewis Research Center,  
National Aeronautics and Space Administration,  
Cleveland, Ohio, March 8, 1965.

PROGRAM PRINTOUT

GAS, LIQUID, LIQUID METAL RADIATOR

NOTE - THE COMPUTED TEMP., LAMDA AND DEL Z PROFILES ARE PRINTED OUT BELOW

THE OUTPUT IS FOR TURBULENT FLOW EQUAS., UNLESS OTHERWISE STATED

INPUT DATA - GAS

0.16400E 05 0.53000E 00 0.19600E-04 0.20000E 02 0.12400E-00 0.96700E 03 0.53600E 03 0.38700E 02  
 C.95000E 03 C.16900E 03 0.16900E 03 0.16900E 03 0.64000E-01 0.80000E-02 0.11100E-03 0.37000E-05  
 C.40000E 03 C.1C500E 03 0.17130E-08 0.53000E-10 0.13400E 01 0.66667E 00 0.44000E-00 0.98400E 05  
 C.90000E 00 1.00000E 00 0.36500E 03 0.90000E 00 0.17500E 01 1.00000E-04 1.00000E-02  
 1.00000E 01 1.00000E-02 -0. -0.

DI= 0.625000E-01 FLK= C.600000E 01 FLAMRD= 1.000000E 00 597.41 570.79 543.17 514.64  
 739.03 717.10 694.71 671.77 647.61 622.87 0.5282 0.4607 0.3970 0.3377  
 1.0000 0.5136 C.8307 0.7511 0.6729 0.5987 1.6873 2.0684 2.6252 3.5166  
 0.7311 C.8146 C.9172 1.0452 1.2030 1.4092 BIGN= 0.390000E 02 WT= 0.491592E 03  
 AH= 0.542371E 02 DNE= 0.437958E-00 DIW= 0.395752E-00 HI= 0.250773E-02 AT= 0.193898E 03  
 WH= C.178913E 03 W= 0.735039E 03 ZBIG= 0.160179E 02 UHEIP= 0.558180E 01  
 ZN= 0.610571E 03 DELTA= 0.167927E-01 DIAT= 0.101085E-00 PANEL= 0.432039E 03  
 TLIT= 0.123738E-01 REYN= 0.141249E 05 WF= 0.645332E 02  
 AHT= 0.279532E-00  
 VNE= 0.174492E 03

INPUT DATA - LIQUID

0.16400E 05 0.14245E 01 0.36500E-02 0.50000E 02 0.39000E-00 0.70600E 03 0.67000E 03 0.69500E 02  
 0.16900E 03 0.16900E 03 0.16900E 03 0.14400E 04 0.28800E 03 0.11000E 03 0.22600E-04 0.40000E 03  
 0.10500E 03 0.17130E-C8 0.53000E-10 0.13400E 01 0.66667E 00 0.44000E-00 0.98400E 05 0.90000E 00  
 1.00000E 00 0.36500E 03 C.90000E 00 0.17500E 01 1.00000E-04 1.00000E-02 1.00000E 01  
 1.00000E 00 -0. -0.

DI= 0.625000E-01 FLK= C.600000E 01 FLAMRD= 1.000000E 00 668.26 664.94 661.61 658.27  
 688.20 684.88 681.56 678.24 674.91 671.59 18.0583 18.4074 19.1403  
 16.1786 16.4682 16.7668 17.0747 17.3923 17.7200 WT= 0.194143E 03 WF= 0.216968E 02  
 WLQ= C.284288E 03 WLQH= 0.763420E 00 WLQI= 0.676851E 02 DELTA= 0.130594E-01 HIGN= 0.200000E 01  
 DIAT= 0.936189E-C1 PANEL= 0.207831E C3 UHEIP= 0.524863E 02 TLIT= 0.864825E-02 REYN= 0.397530E 04  
 ZN= 0.317138E C3 HI= 0.218387E-01 VNE= 0.334040E 01

\*01\* UNITS, ECF.

## REFERENCES

1. Saule, Arthur V.; Krebs, Richard P.; and Auer, Bruce M.: Design Analysis and general Characteristics of Flat-Plate Central-Fin-Tube Sensible-Heat Space Radiators. NASA TN D-2839, 1965.
2. Kalaba, Robert: On Nonlinear Differential Equations, the Maximum Operation, and Monotone Convergence. J. of Math. and Mech., vol. 8, no. 4, 1959, pp. 519-574.